

IN THE ABSTRACT:

Please delete the abstract and substitute the following abstract therefore:

An apparatus for aligning a gas turbine engine blade including an airfoil and a dovetail is provided. The apparatus includes at least one locator pin configured to engage a serration formed on the blade dovetail, a locator block supporting said locator pin where said locator block comprises at least one groove sized to receive said locator pin therein, and a slide block assembly for engaging a dovetail surface opposite the serration where said slide block assembly is configured to position the blade dovetail against said locator pin. The apparatus also includes a base member comprising a platform comprising an end plate, a bottom surface, and an opposite upper surface for supporting said slide block assembly and said locator block, said slide block assembly is slidably coupled to said platform upper surface, said platform upper surface defining a slotted opening extending through said bottom surface, said opening configured to receive the blade airfoil.